

4/1

Mass Spring Damper

$$m\ddot{y} + \gamma\dot{y} + ky = v$$

$$\mathcal{L}[m\ddot{y} + \gamma\dot{y} + ky] = \mathcal{L}[v]$$

$$y = y(t)$$

$$v = v(t)$$

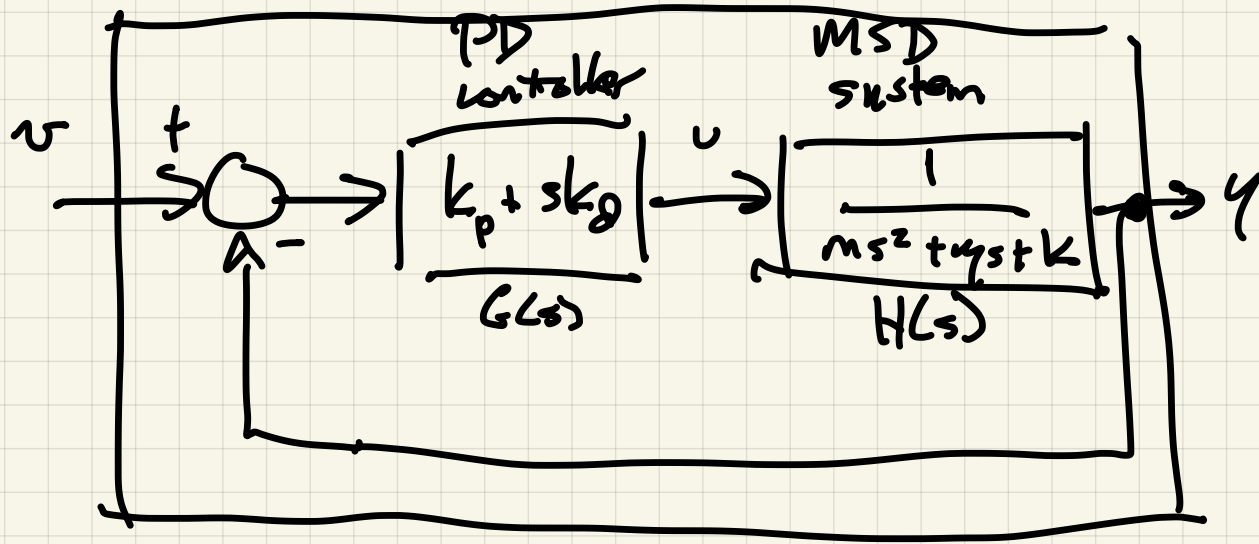
$$ms^2 Y(s) + \gamma s Y(s) + k Y(s) = U(s)$$

$$(ms^2 + \gamma s + k) Y(s) = U(s)$$

$$Y(s) = \frac{1}{\underbrace{ms^2 + \gamma s + k}_{H(s)}} U(s)$$

$$H(s) = \frac{Y(s)}{U(s)}$$

PD



$m > n$
not causal



$$T_{cl} = \frac{G(s)H(s)}{1 + G(s)H(s)}$$

$$G(s) = k_D s + k_p = \frac{k_D s + k_p}{1}$$

$$= \frac{(k_D s + k_p) \left(\frac{1}{m s^2 + \gamma s + k} \right)}{1 + (k_D s + k_p) \left(\frac{1}{m s^2 + \gamma s + k} \right)}$$

$$= \frac{k_D s + k_p}{m s^2 + \gamma s + k + k_D s + k_p} = \frac{k_D s + k_p}{m s^2 + (\gamma + k_D) s + (k + k_p)}$$

Steady State (step response)

$$y_{ss} = \lim_{t \rightarrow \infty} y(t) = \lim_{s \rightarrow 0} s H(s) \frac{1}{s} = \frac{k_p}{k + k_p} < 1$$

PID

$$\text{controller } G(s) = k_p + k_d s + \frac{k_i}{s}$$
$$= \frac{k_d s^2 + k_p s + k_i}{s}$$

still not causal!

$$\begin{aligned} \overline{T}_L(s) &= \frac{G H}{1 + G H} = \frac{k_d s^2 + k_p s + k_i}{s} \cdot \frac{1}{m s^2 + \gamma s + k} \\ &= \frac{1 + \frac{k_d s^2 + k_p s + k_i}{s}}{m s^2 + \gamma s + k} \cdot \frac{1}{m s^2 + \gamma s + k} \\ &= \frac{k_d s^2 + k_p s + k_i}{s(m s^2 + \gamma s + k) + k_d s^2 + k_p s + k_i} \\ &= \frac{k_d s^2 + k_p s + k_i}{m s^3 + (\gamma + k_d) s^2 + (k + k_p) s + k_i} \end{aligned}$$

$$y_{ss} = \lim_{s \rightarrow 0} s \cdot H(s) \frac{1}{s} = \frac{k_i}{k_i} = 1 !!$$

Causality:

$$\dot{x} = Ax + Bu$$
$$y = Cx$$

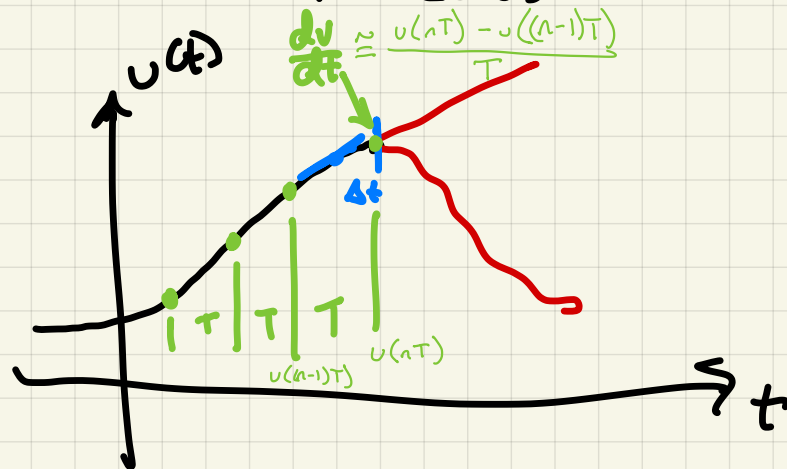
$m < n$ strictly causal \Rightarrow ss realization (A, B, C)

$m \leq n$ causal \Rightarrow ss realization (A, B, C, D)

$$\dot{x} = Ax + Bu$$

$$y = Cx + Du$$

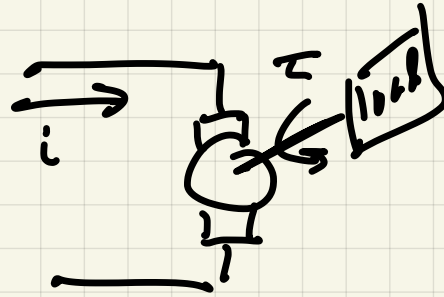
$m > n$ noncausal \Rightarrow no ss representation



Mobile Robots: levels of control

Brushed DC Motor

assume current is input



torque constant

$$\tau = K_T i$$

motor velocity: ω

$$i \rightarrow \boxed{H(s)} \rightarrow \omega$$

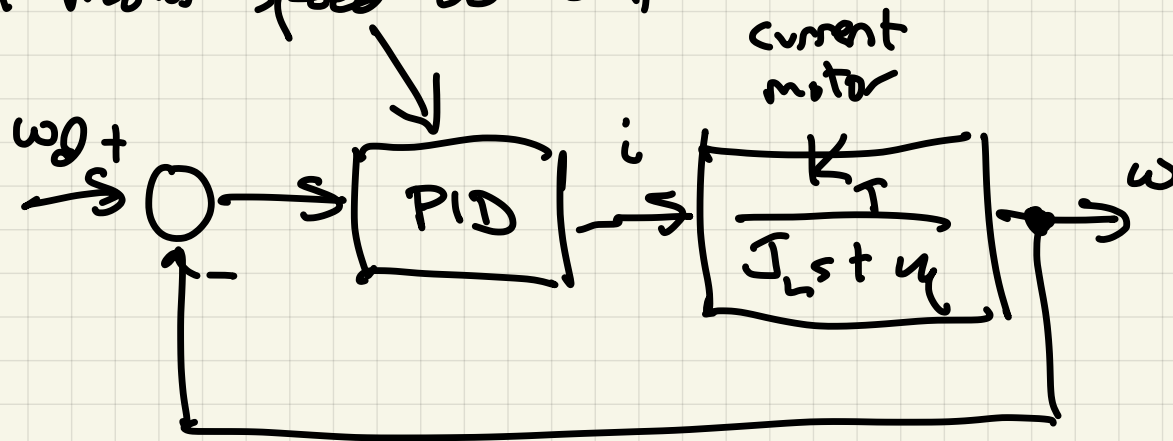
ODE (newton's law)

$$\tau = J_L \dot{\omega} + \eta \omega = K_T i$$

moment of inertia of load

$$H(s) = \frac{K_T}{J_L s + \eta}$$

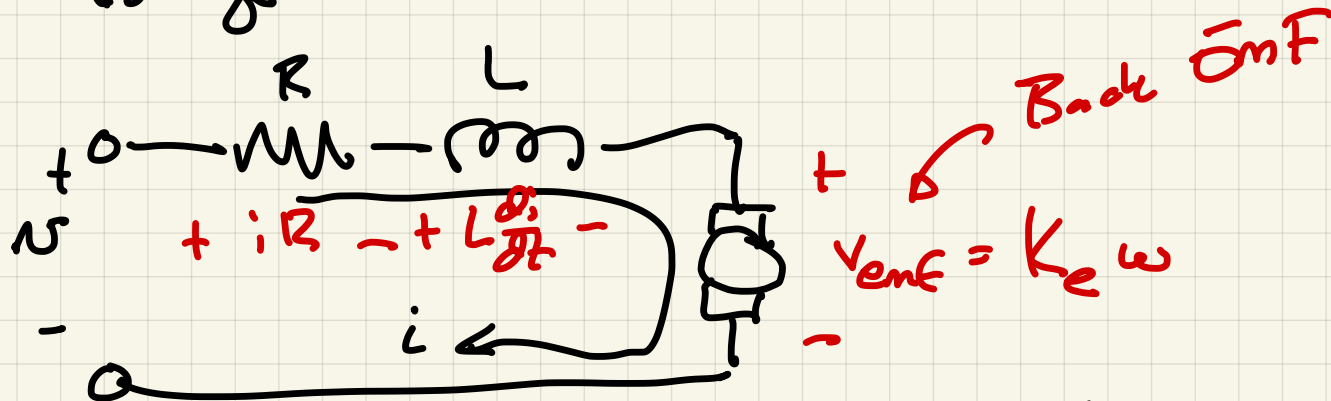
current motor speed controller



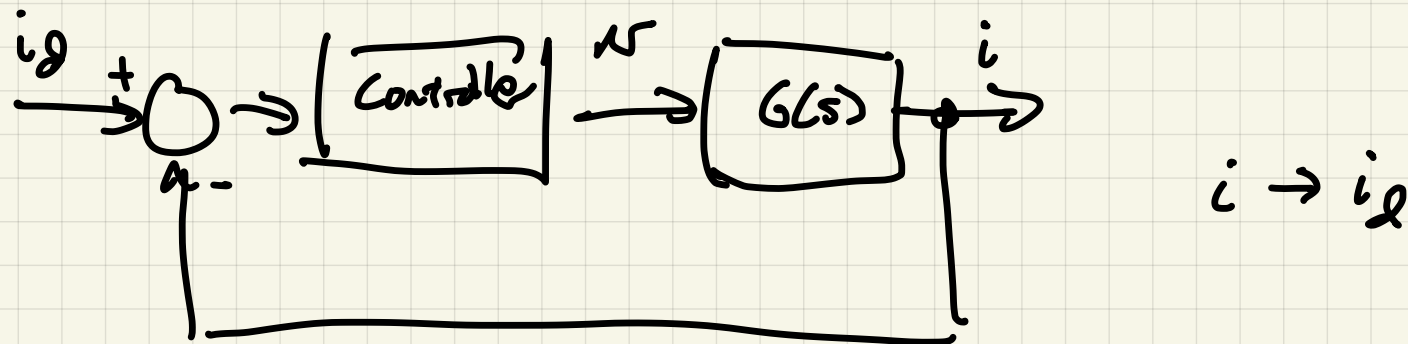
$\omega \rightarrow \omega_d$

But, we don't really control motor current, so it is not a good input

input as voltage v



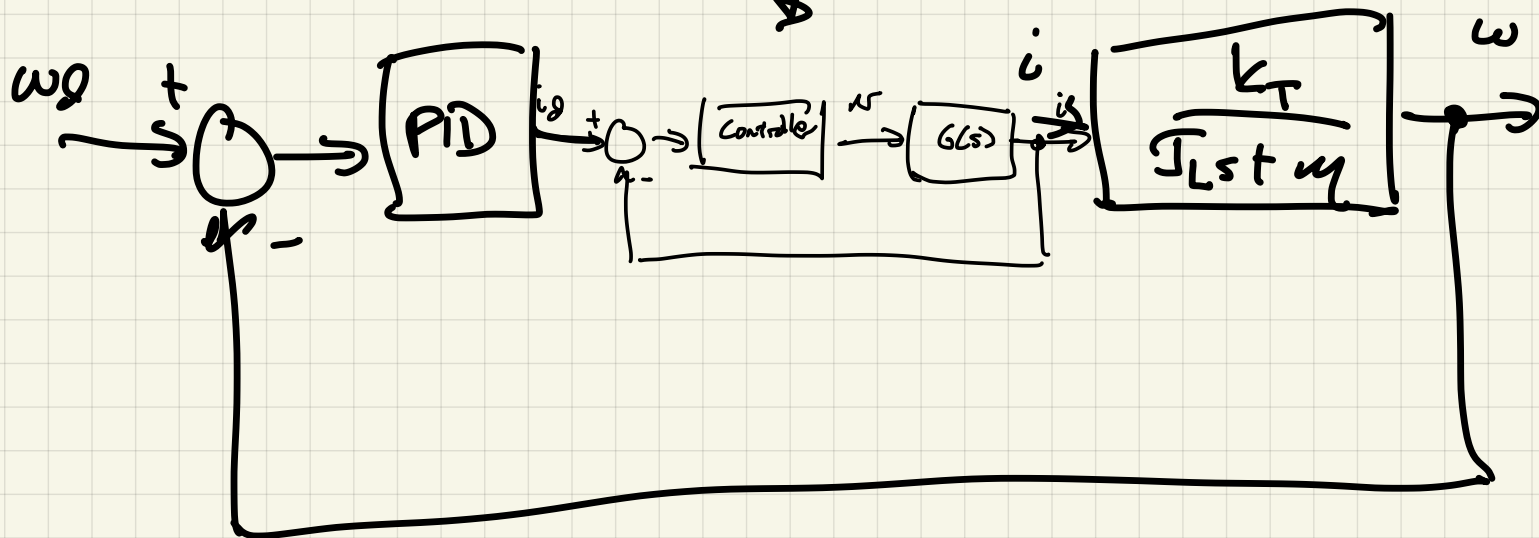
consider i as output $v \rightarrow \boxed{G(s)} \rightarrow i$

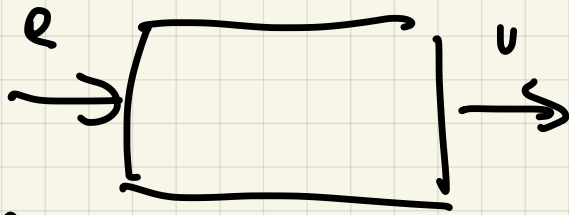


$i \rightarrow i_l$

back to current as input

voltage to current amplifier





$$\mathcal{L}\left[u = k_p e + k_d \frac{de}{dt}\right]$$

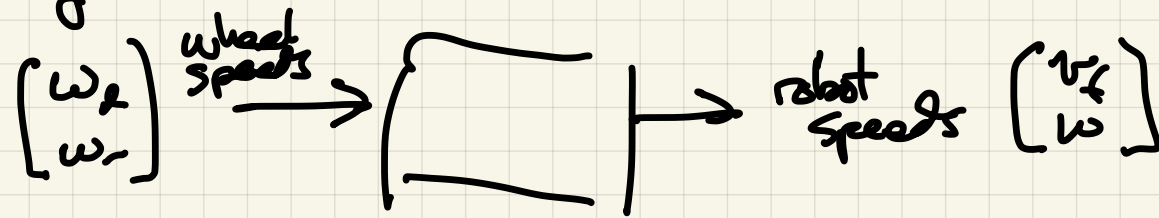
$$U(s) = k_p \bar{E} + s k_d \bar{E}$$

$$= \underbrace{(k_p + s k_d)}_{G(s)} \bar{E}$$

4/6 Differential Drive Robot



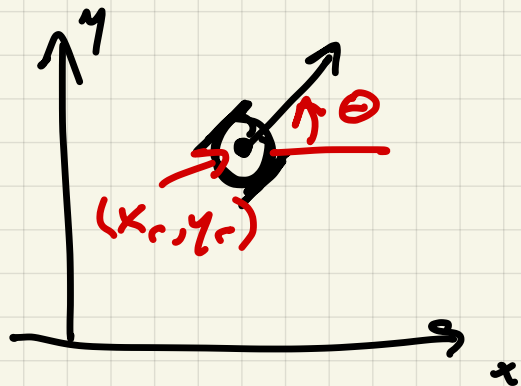
Model: given wheel speeds, what are the robot



Kinematic Model (as opposed to a dynamic model)

↳ the study of motion without regard to the forces that cause it

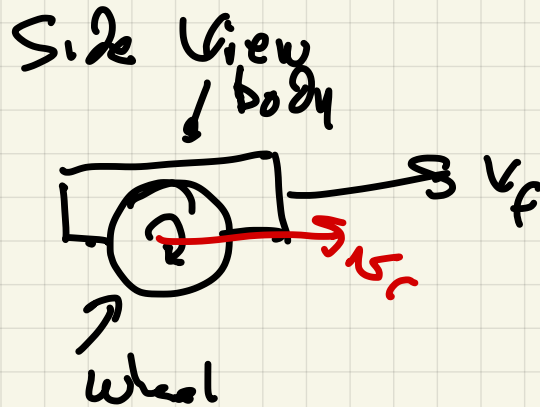
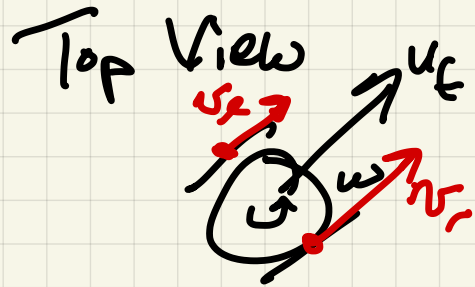
Robot Pose: position and orientation



$$q = \begin{bmatrix} x_r \\ y_r \\ \theta \end{bmatrix} = x \text{ pose configuration}$$

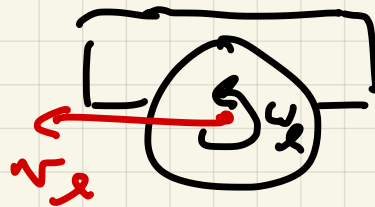
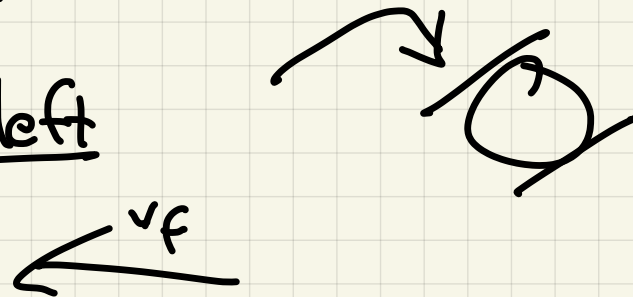
$$\dot{q} = \begin{bmatrix} v_f \\ w \end{bmatrix}$$

Robot Carbons



Intermediate Speed: wheel linear velocities $\begin{bmatrix} v_e \\ v_r \end{bmatrix}$

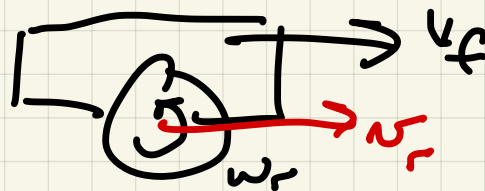
Side view from left



R = wheel radius

$$v_e = R\omega_e$$

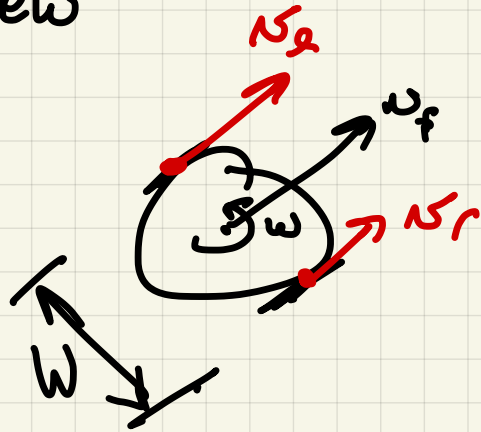
Side view from right



$$v_r = -R\omega_r$$

Robot Velocities

Top View



$$v_f = \frac{v_l + v_r}{2}$$

W = wheelbase

$$\omega = \frac{v_r - v_l}{W}$$

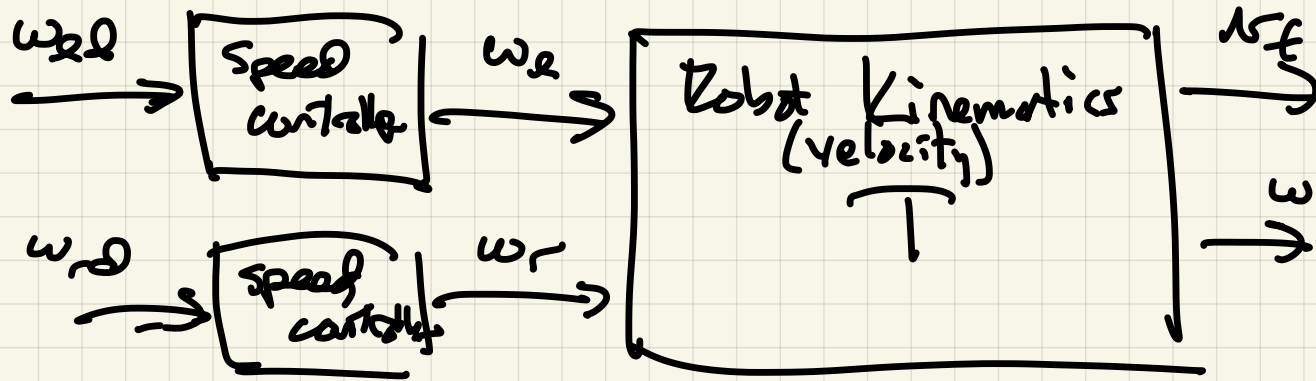
Combine:

$$v_f = \frac{-\omega_r R + \omega_l R}{2} = -\frac{R}{2} \omega_r + \frac{R}{2} \omega_l$$

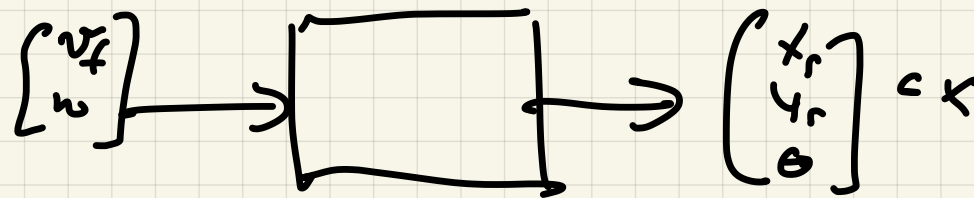
$$\omega = \frac{-\omega_r R - \omega_l R}{W} = -\frac{R}{W} \omega_r - \frac{R}{W} \omega_l$$

$$\begin{bmatrix} v_f \\ \omega \end{bmatrix} = \begin{bmatrix} \frac{R}{2} & -\frac{R}{2} \\ \frac{R}{W} & -\frac{R}{W} \end{bmatrix} \begin{bmatrix} \omega_l \\ \omega_r \end{bmatrix}$$

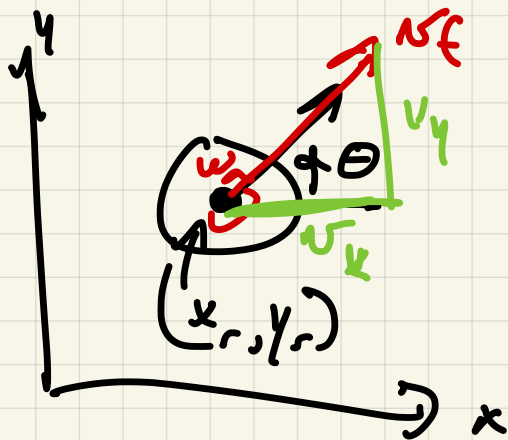
$$\begin{bmatrix} \omega_l \\ \omega_r \end{bmatrix} \rightarrow \boxed{T} \rightarrow \begin{bmatrix} v_f \\ \omega \end{bmatrix}$$



Another Model takes robot velocities to robot pose



Still kinematic



$$\dot{x} = \begin{bmatrix} v_f \cos \theta \\ v_f \sin \theta \\ \omega \end{bmatrix}$$

$$\text{input } \begin{bmatrix} v_f \\ \omega \end{bmatrix} = u$$

$$\dot{x} = f(x, u)$$

$$\text{state } x = \begin{bmatrix} x_r \\ y_r \\ \theta \end{bmatrix}$$

$$\dot{x} = \begin{bmatrix} v_f \cos \theta \\ v_f \sin \theta \\ \omega \end{bmatrix} = \begin{bmatrix} v_1 \cos \kappa_3 \\ v_1 \sin \kappa_3 \\ v_2 \end{bmatrix}$$

$$= \underbrace{\begin{bmatrix} \cos \kappa_3 \\ \sin \kappa_3 \\ 0 \end{bmatrix}}_{g_1} v_1 + \underbrace{\begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}}_{g_2} v_2$$

drift free affine state space model

$$= \sum_{i=1}^m g_i v_i$$

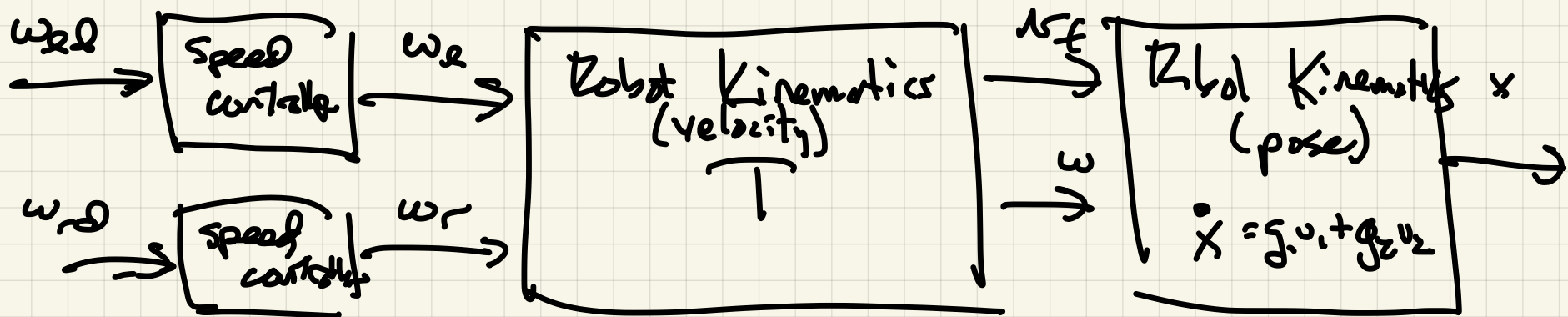
g_1, g_2 control vector fields

Drift Free when $v = 0$ the \dot{x} also is 0
every state is an equilibrium point!

(affine system with drift)

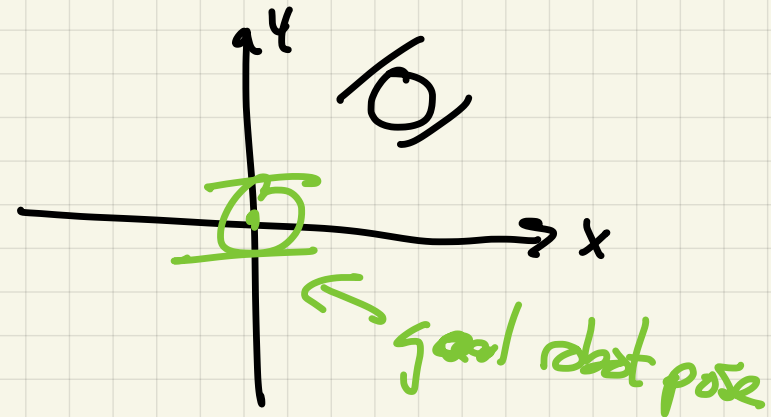
$$\dot{x} = f(x) + g_1(x) v_1 + g_2(x) v_2 + \dots + g_m(x) v_m$$

↑
drift vector field



First Try to Linearize

$$\dot{x} = f(k, v) = \begin{bmatrix} v_1 \cos k_3 \\ v_1 \sin k_3 \\ v_2 \end{bmatrix}$$



$$\dot{x} = \left. \frac{\partial f}{\partial x} \right|_{k=0, v=0} x + \left. \frac{\partial f}{\partial v} \right|_{k=0, v=0} v$$

$$= \begin{bmatrix} 0 & 0 & -v_1 \sin k_3 \\ 0 & 0 & v_1 \cos k_3 \\ 0 & 0 & 0 \end{bmatrix} x - \begin{bmatrix} \cos k_3 & 0 \\ \sin k_3 & 0 \\ 0 & 1 \end{bmatrix} v$$

$k=0, v=0$ $k=0, v=0$

$$\dot{x} = \begin{bmatrix} 0 & 0 & 0 \\ v & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} x + \begin{bmatrix} 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{bmatrix} v$$

$$\dot{x} = \begin{pmatrix} 0 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{pmatrix} x + \begin{pmatrix} 1 & 0 \\ 0 & 0 \\ 0 & 1 \end{pmatrix} u$$

Check controllability

$$Q = [B \quad AB \quad A^2B]$$

$$= \begin{pmatrix} 1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & 0 & 0 \end{pmatrix} \Rightarrow \text{rank} = 2 \Rightarrow \text{not controllable}$$

Original nonlinear system, is it controllable? Yes!

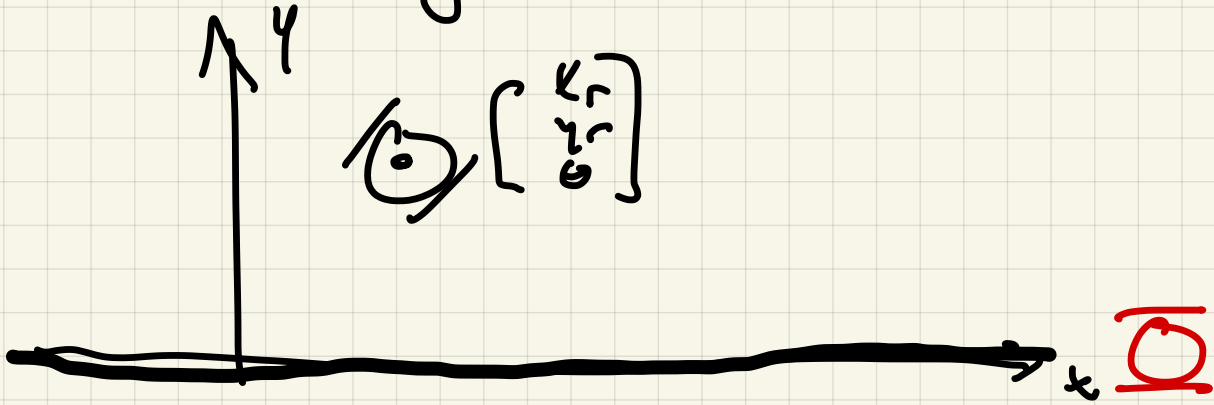
But - linearization is not controllable!

Ans - dimension of control space is 2
 dimension of state space is 3
 $2 < 3$

"not altogether lawful"

\Rightarrow there is a nonholonomic constraint.
 \Rightarrow the system is under actuated,

Line following



let v_f be constant

new state $z = \begin{bmatrix} x_r \\ 0 \end{bmatrix}$

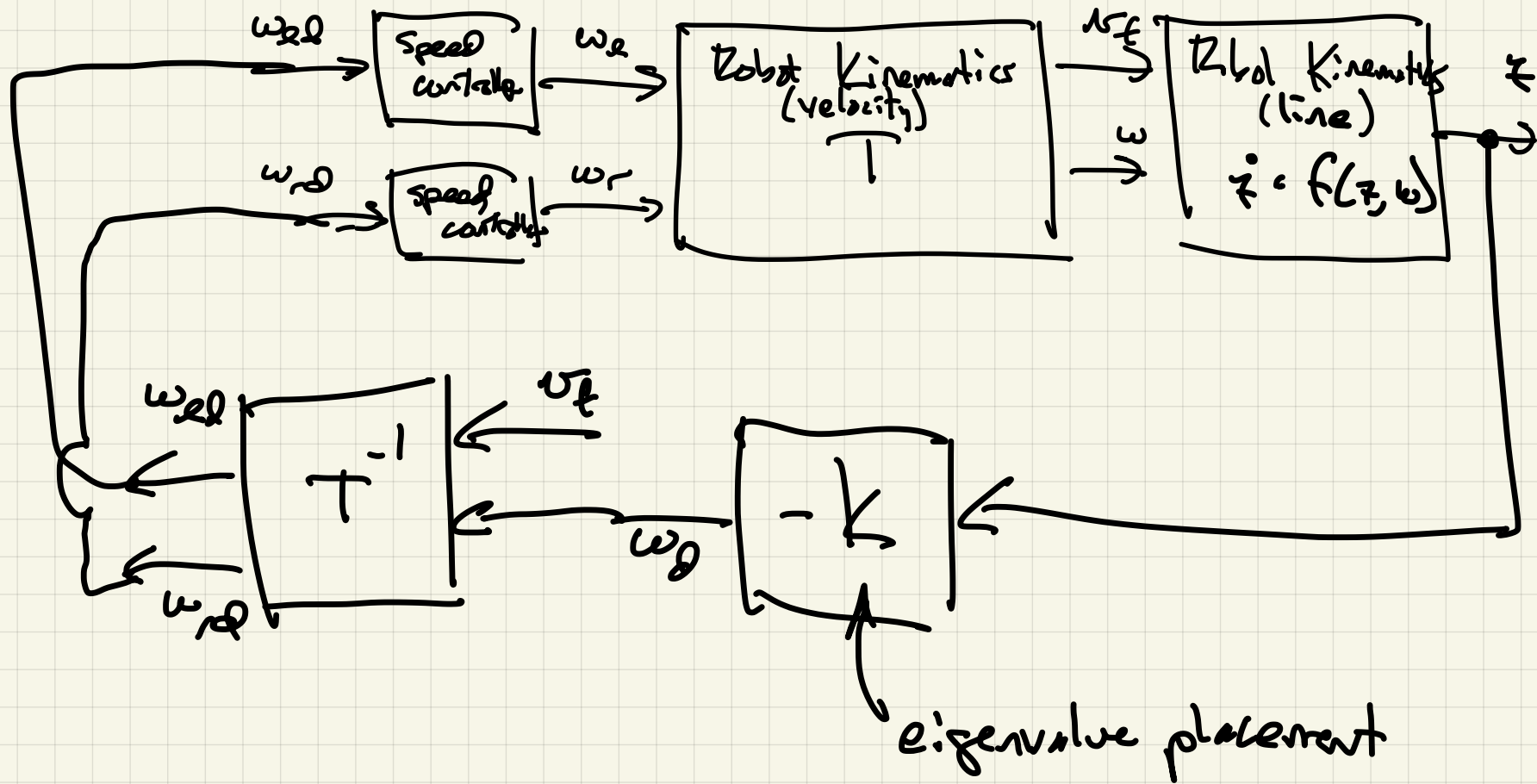
$$\dot{x} = f(x, u) = \begin{bmatrix} v_f \cos k_3 \\ v_f \sin k_3 \\ v_f \end{bmatrix}$$

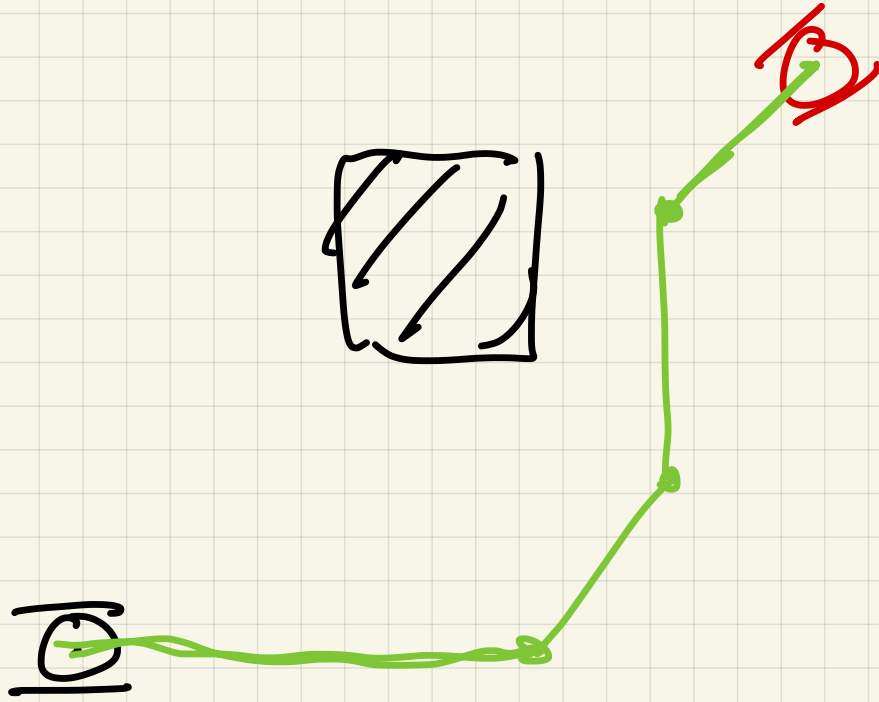
$$\dot{z} = \begin{bmatrix} v_f \sin z_2 \\ w \end{bmatrix}$$

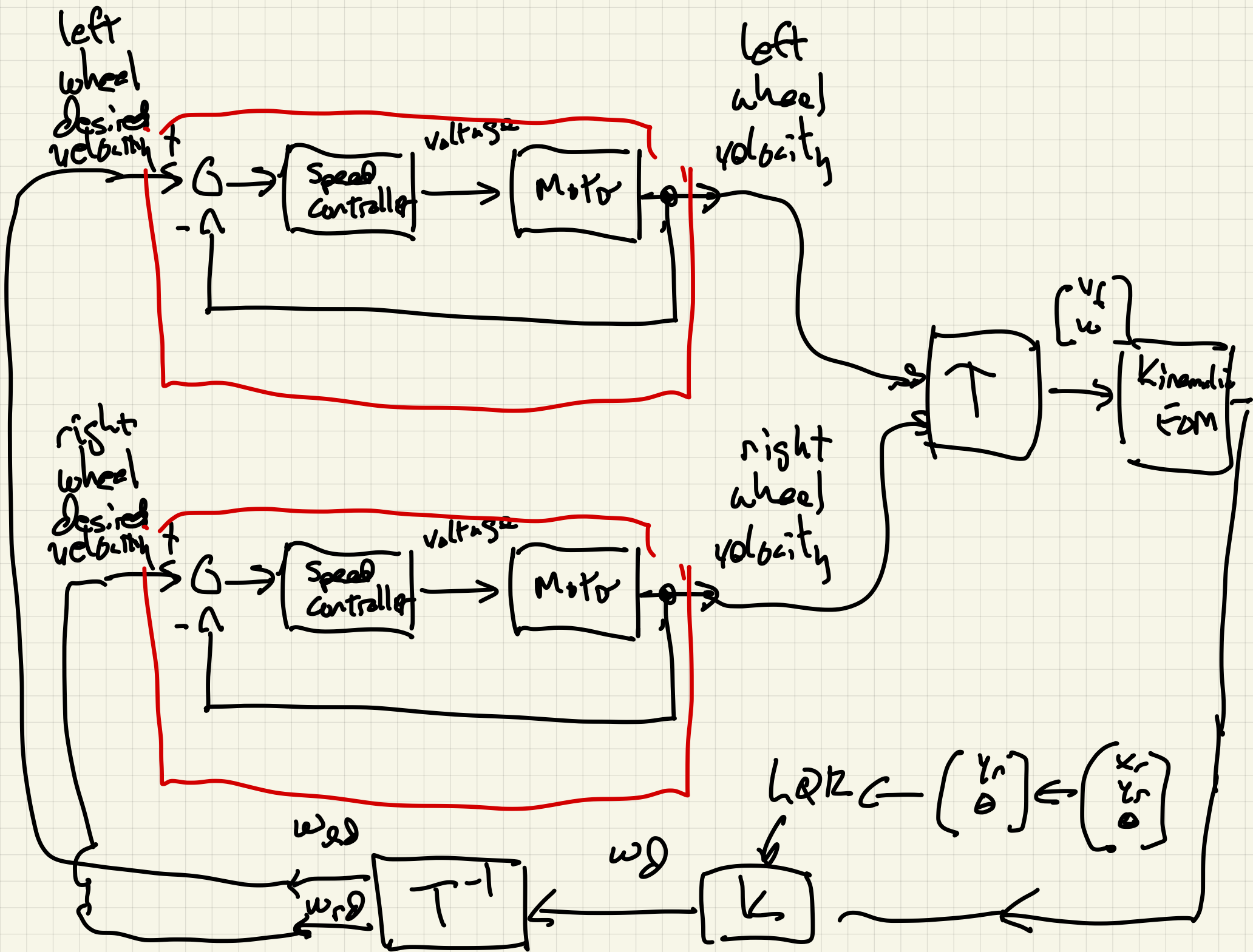
linearize about $z=0$

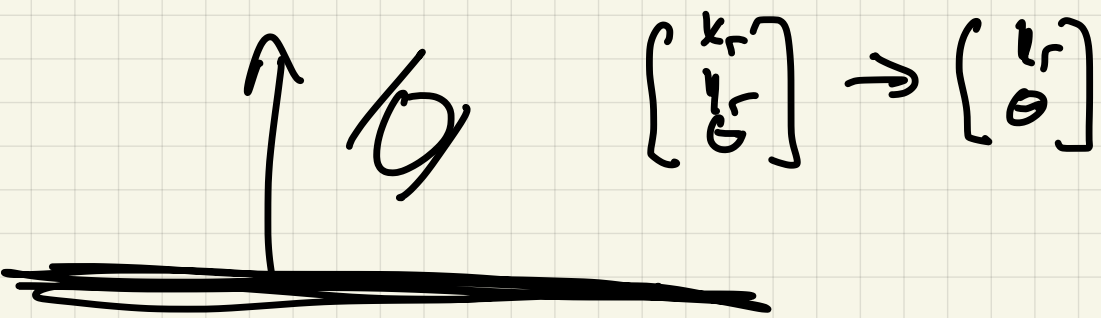
$$\dot{z} = \begin{bmatrix} 0 & v_f \\ 0 & 0 \end{bmatrix} z + \begin{bmatrix} 0 \\ 1 \end{bmatrix} w$$

$$Q = \begin{bmatrix} 0 & v_f \\ 1 & 0 \end{bmatrix} \quad \text{rank} = 2 \Rightarrow \text{controllable!}$$

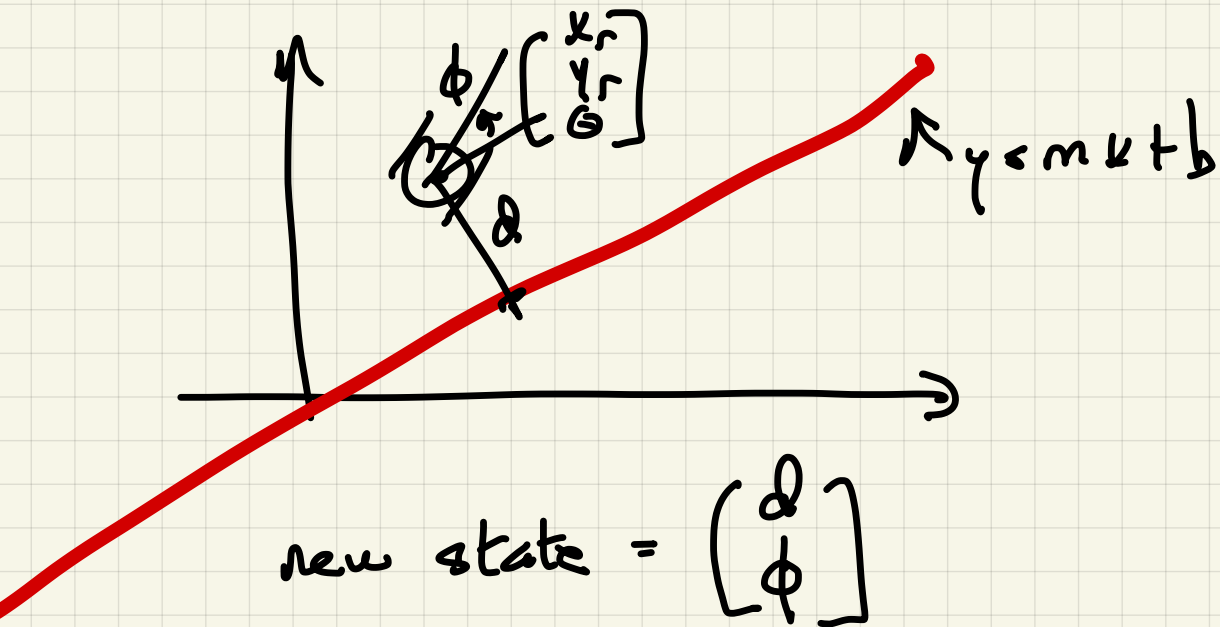


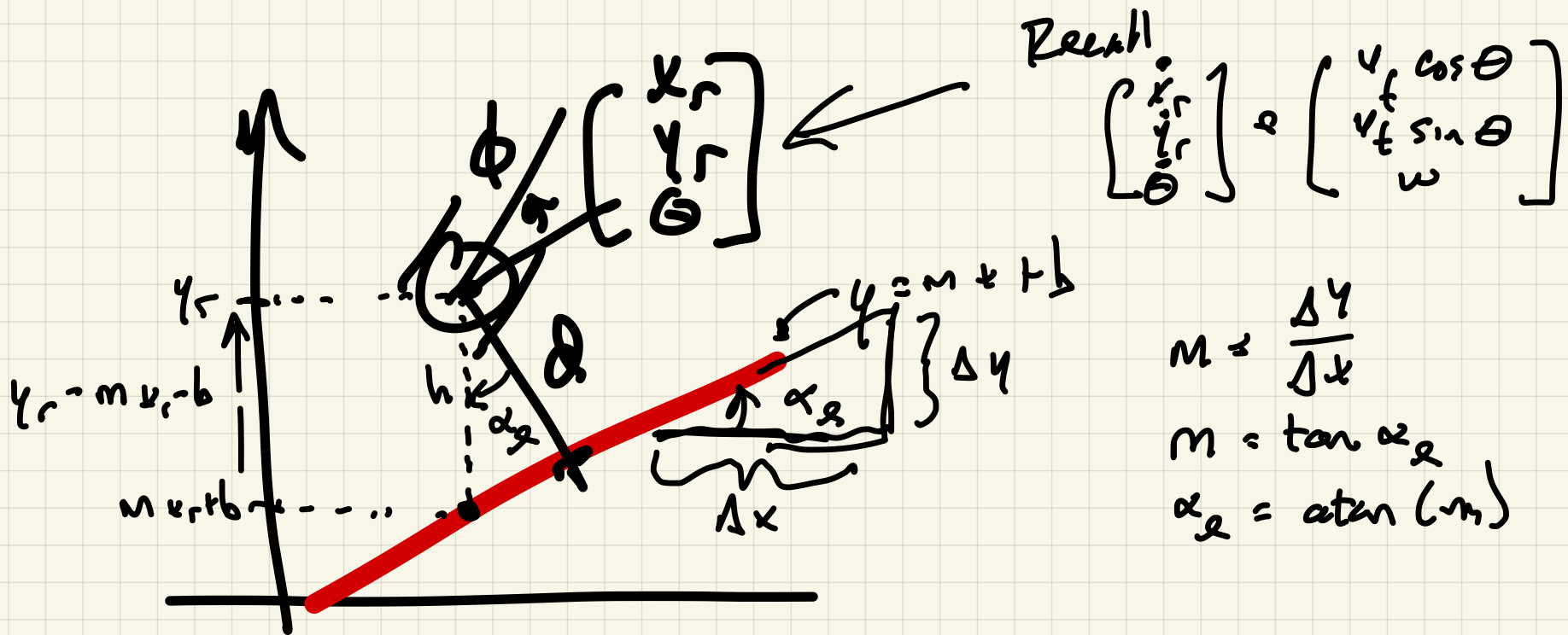






Different Line





$$h = y_r - mx_r - b \Rightarrow d = (y_r - mx_r - b) \cos \alpha_e$$

$$\phi = \theta - \alpha_e$$

goal: $\dot{z} = f(z, v)$

$$z = \begin{bmatrix} d \\ \phi \end{bmatrix}$$

$$\dot{z} = \begin{bmatrix} \dot{d} \\ \dot{\phi} \end{bmatrix}$$

\Rightarrow

$$\dot{d} = \cos \alpha_e (\dot{y}_r - m \dot{x}_r)$$

$$= \cos \alpha_e (v_f \sin \theta - m v_f \cos \theta)$$

$$= v_f \cos \alpha_e (\sin(\phi + \alpha_e) - m \cos(\phi + \alpha_e))$$

$$\dot{d} = v_f \cos \alpha_e (\sin(\phi + \alpha_e) - m \cos(\phi + \alpha_e))$$

$$\dot{\phi} = \dot{\Theta} = \omega$$

Intuitively $z = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$ shall be an equilibrium point

$$\text{for input} = \begin{bmatrix} v_f \\ 0 \end{bmatrix}$$

set $\dot{d} = \dot{\phi} = \omega = 0$, $v_f = \text{constant}$

$$\dot{\phi} = 0 \quad \checkmark$$

Recall :

$$\sin(\phi + \alpha_e) = \sin \phi \cos \alpha_e + \cos \phi \sin \alpha_e$$

$$\cos(\phi + \alpha_e) = \cos \phi \cos \alpha_e - \sin \phi \sin \alpha_e$$

$$\begin{aligned} \Rightarrow \dot{d} &= v_f \cos \alpha_e \left(\cancel{\sin \phi} \cos \alpha_e + \cos \phi \sin \alpha_e - m \cos \phi \cos \alpha_e + m \cancel{\sin \phi} \sin \alpha_e \right) \\ &= v_f \cos \alpha_e (\sin \alpha_e - m \cos \alpha_e) \end{aligned}$$

$$\dot{d} = v_f \cos \alpha_e (\sin \alpha_e - m \cos \alpha_e)$$

$$m = \frac{\Delta k}{\Delta y} \text{ s } \tan \alpha_e = \frac{\sin \alpha_e}{\cos \alpha_e}$$

$$\dot{d} = v_f \cos \alpha_e (\cancel{\sin \alpha_e} - \frac{\cancel{\sin \alpha_e}}{\cancel{\cos \alpha_e}} \cancel{\cos \alpha_e}) = 0 !!$$

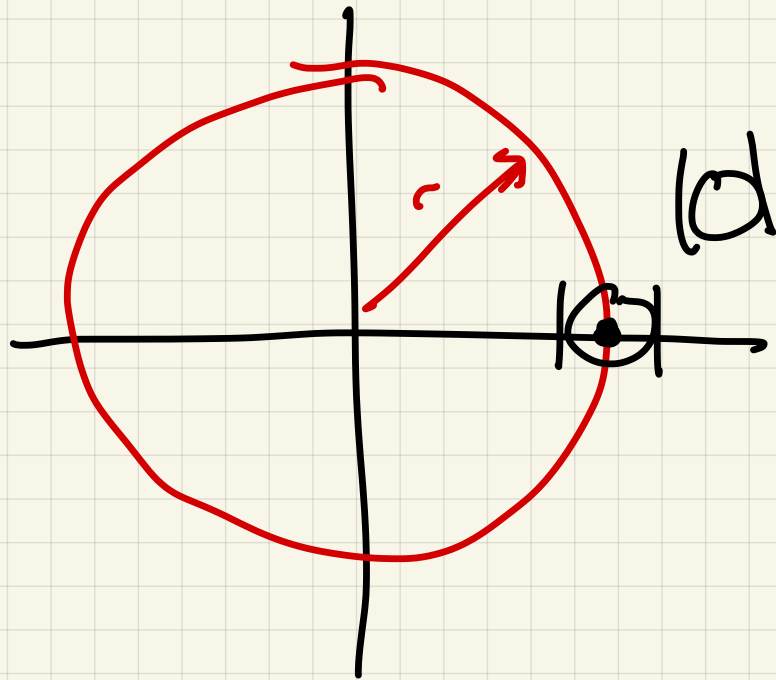
$$\dot{d} = v_f \cos \alpha_e (\sin(\phi + \alpha_e) - m \cos(\phi + \alpha_e))$$

$$\dot{\phi} = \dot{\Theta} = \omega$$

Linearize about $\xi_e = \begin{bmatrix} 0 \\ 0 \end{bmatrix}$, check controllability,
do state feedback.

What if trajectory is not a line?

Try a circle



Is there an equilibrium trajectory?

Start at

$$x_0 = \begin{bmatrix} r \\ 0 \\ \pi/2 \end{bmatrix}$$

make $v_f \neq 0$ constant

Is there a constant ω that keeps robot on circle?

Yes! what is it?

circumference of circle: $2\pi r$

forward velocity is v_f

distance = rate \times time

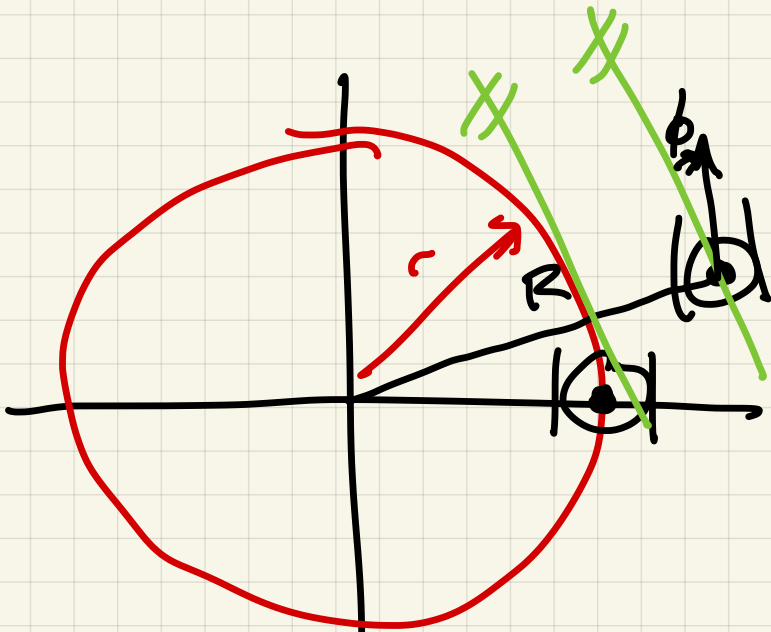
\Rightarrow time to complete the circle is $t_c = \frac{2\pi r}{v_f}$

$$\dot{\theta} = \omega = \frac{2\pi}{2\pi r / v_f} = \frac{v_f}{r}$$

Equilibrium trajectory

initial condition x_0 constant input = $\begin{bmatrix} v_f \\ v_f/r \end{bmatrix}$

$$(R-r) \rightarrow 0$$



$$z = \begin{bmatrix} \overbrace{(R-r)}^{e_r} \\ \phi \end{bmatrix} = \begin{bmatrix} e_r \\ \phi \end{bmatrix}$$

$$\begin{matrix} \dot{e}_r = \dot{r} \\ \dot{\phi} = \dot{\gamma} \end{matrix} f(\gamma, v)$$

Ad Hoc

v_f constant

Conceptually

If robot is outside circle, turn left
inside circle, turn right

P controller?

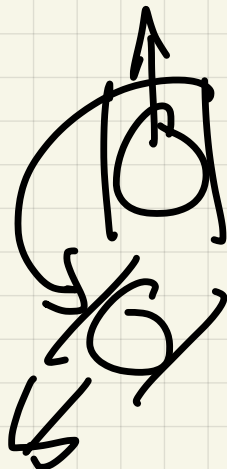
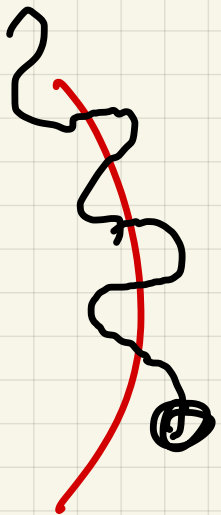
$$\omega = k_p (R - r)$$

tune k_p by trial and error

There might be overshoot

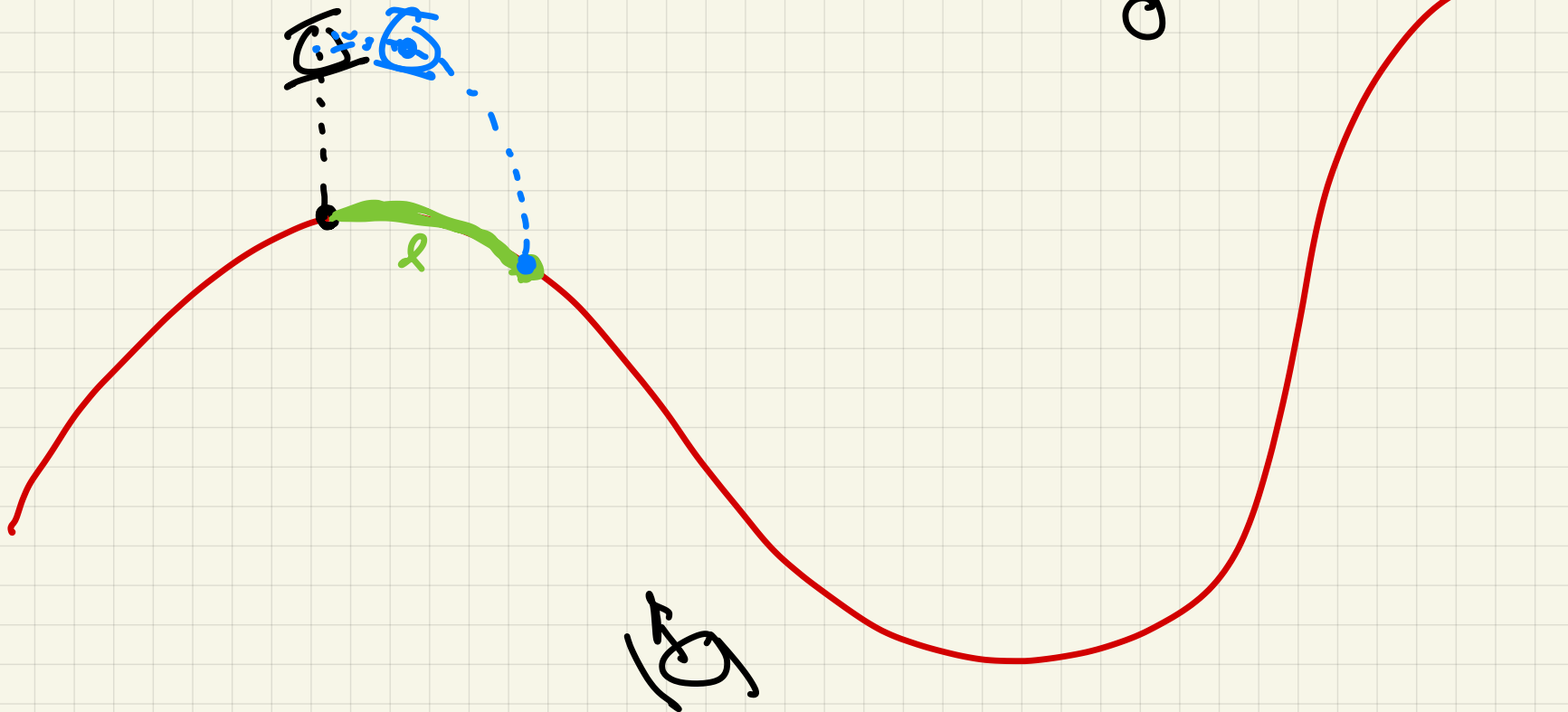
\Rightarrow add a D term

$$\omega = k_p (R - r) + k_d \dot{R}$$

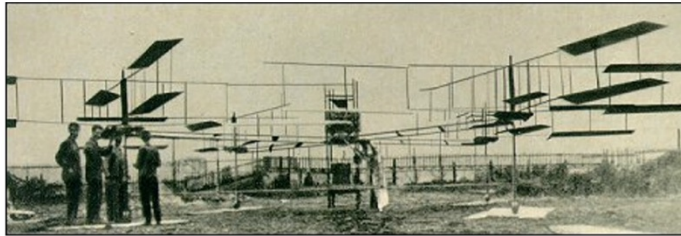


Pure Pursuits

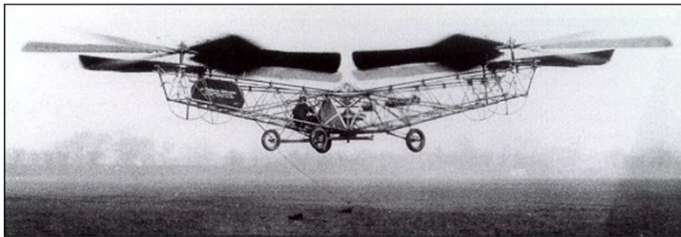
- ① project robot position onto path
- ② look ahead on the path by some distance l
- ③ calculate arc that hits lookahead point
- ④ start following arc



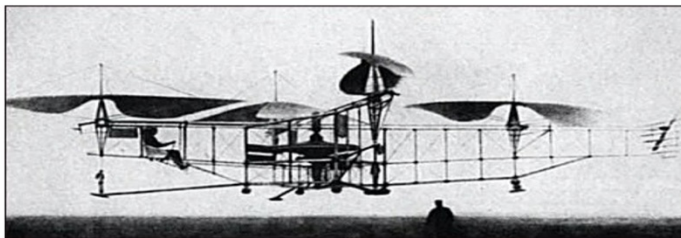
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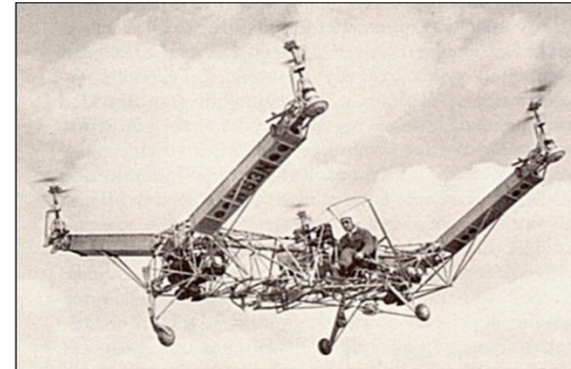
(a)



(b)



(c)



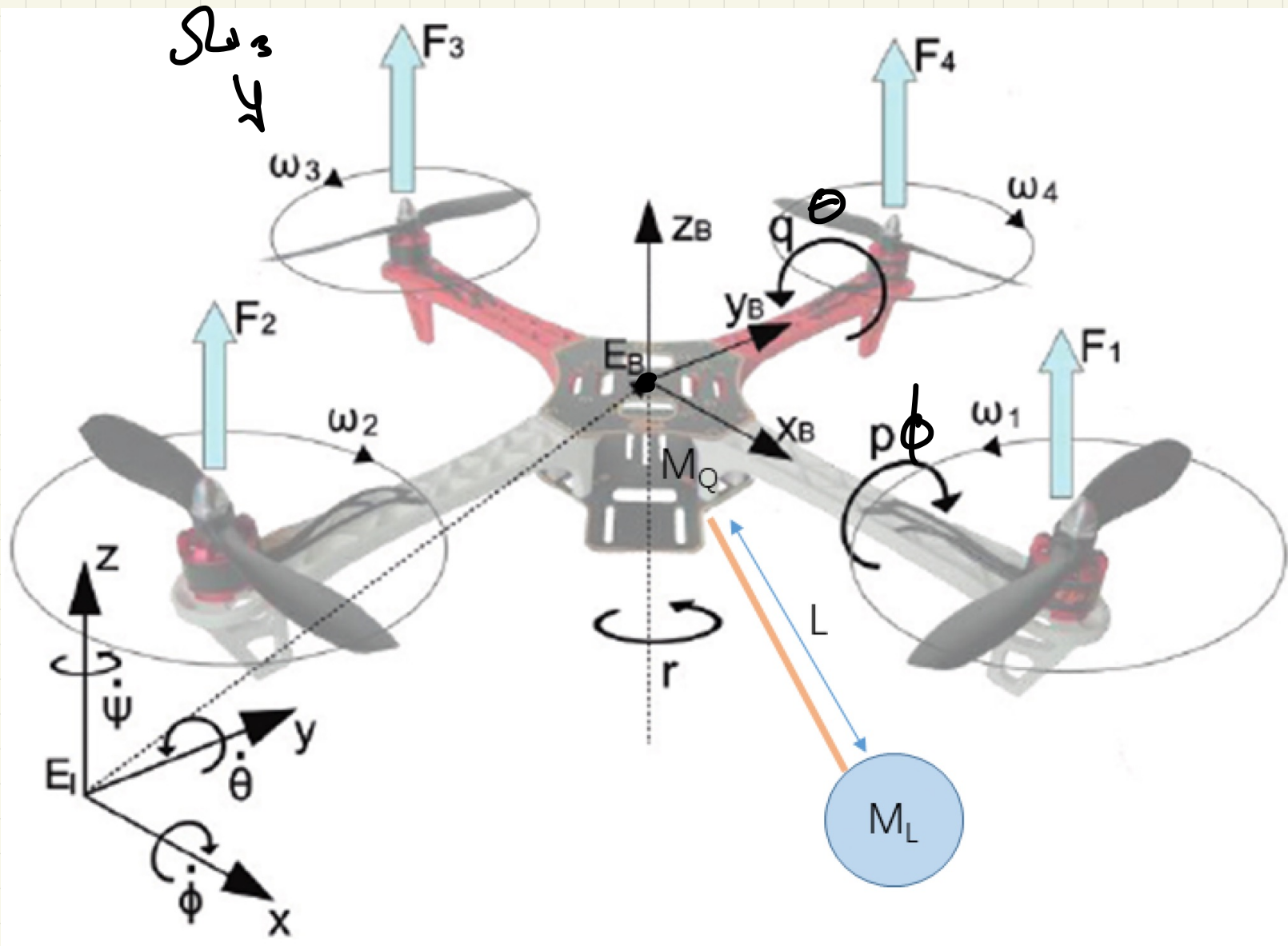
(d)



(e)

Source: Kim, Jinho, S. Andrew Gadsden, and Stephen A. Wilkerson. "A comprehensive survey of control strategies for autonomous quadrotors." *Canadian Journal of Electrical and Computer Engineering* 43, no. 1 (2019): 3-16.

Fig. 3. History of quadrotor: (a) Brèguet-Richet Gyroplane No. 1; (b) Oehmichen No.2; (c) Bothezat helicopter; (d) Convertawings Model A; (e) Curtiss-Wright VZ-7



ψ
 θ
 ϕ
 ω_1
 ω_2
 ω_3
 ω_4
 propeller speeds
 ?

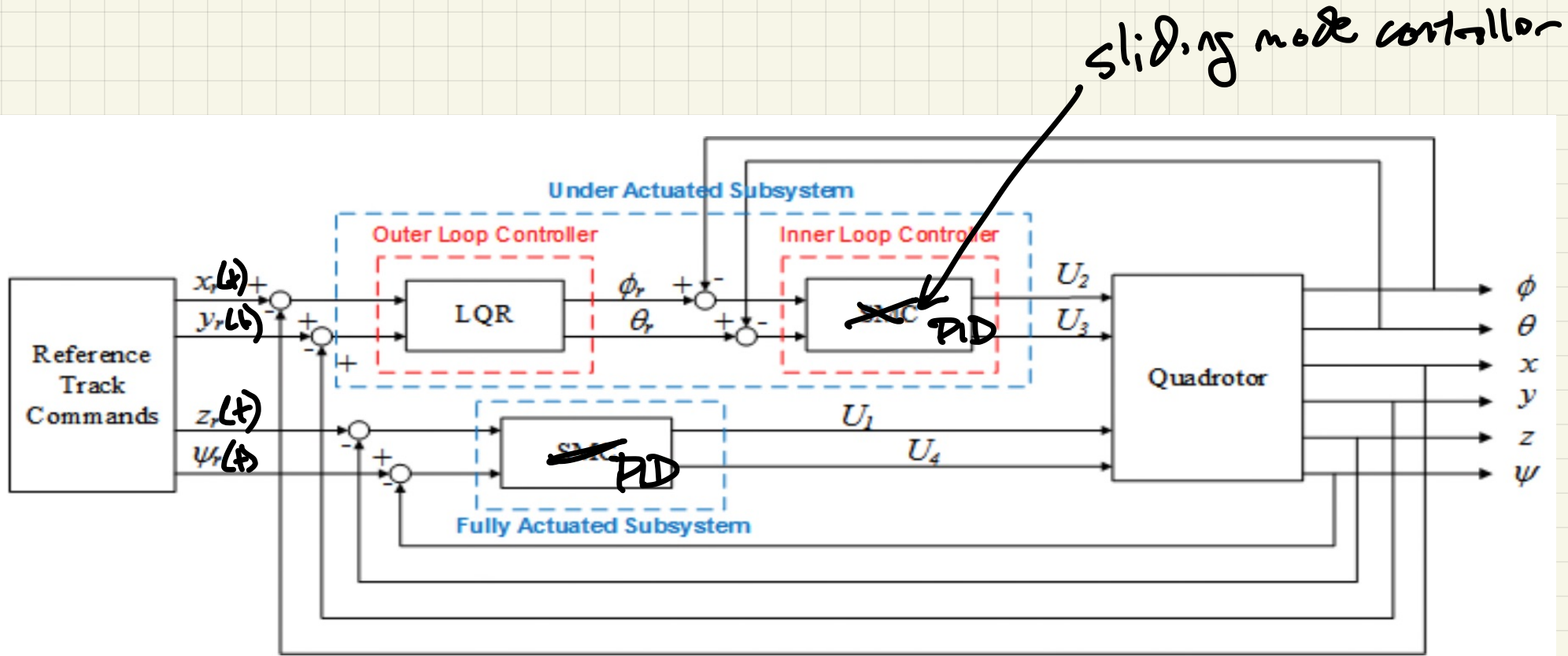
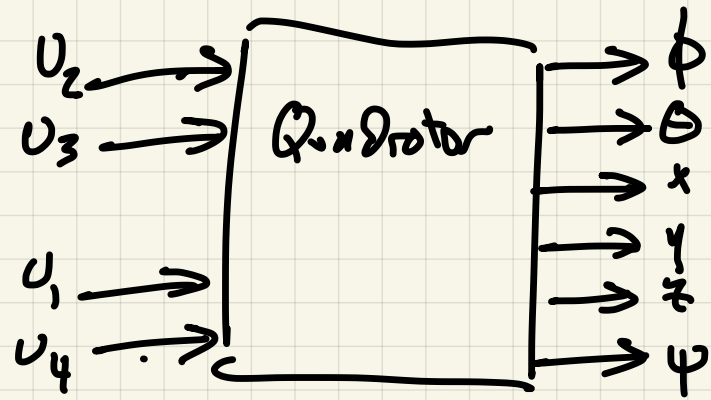
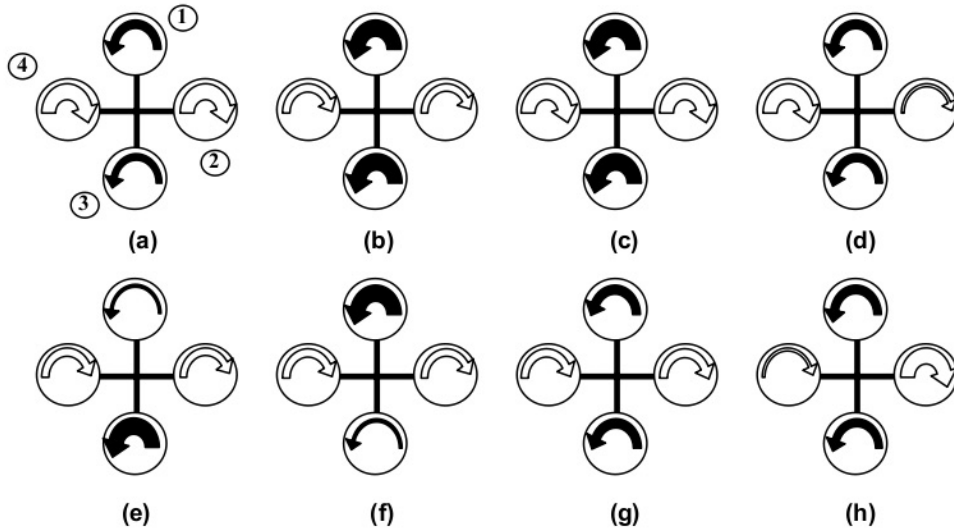


Fig. 2. UAV control system block diagram.

Ghamry, Khaled A., and Youmin Zhang. "Formation control of multiple quadrotors based on leader-follower method." In 2015 International Conference on Unmanned Aircraft Systems (ICUAS), pp. 1037-1042. IEEE, 2015.



U's are kind of like thrust thrust constant
 Single Propeller $T \approx b \omega^2$



- | | | | |
|-----|-------------------------------|-----|---------------------------------|
| (a) | Yaw (anticlockwise direction) | (e) | Pitch (anticlockwise direction) |
| (b) | Yaw (clockwise direction) | (f) | Pitch (clockwise direction) |
| (c) | Take-off or take-up | (g) | Land or take-down |
| (d) | Roll (clockwise direction) | (h) | Roll (anticlockwise direction) |

(a, b) conservation of momentum

$$U_4 = d(-\omega_1^2 + \omega_2^2 - \omega_3^2 + \omega_4^2)$$

(c, g) up and down

$$U_1 = b(\omega_1^2 + \omega_2^2 + \omega_3^2 + \omega_4^2)$$

(e, f) roll, tilt about x axis

$$U_3 = b(\omega_1^2 - \omega_3^2)$$

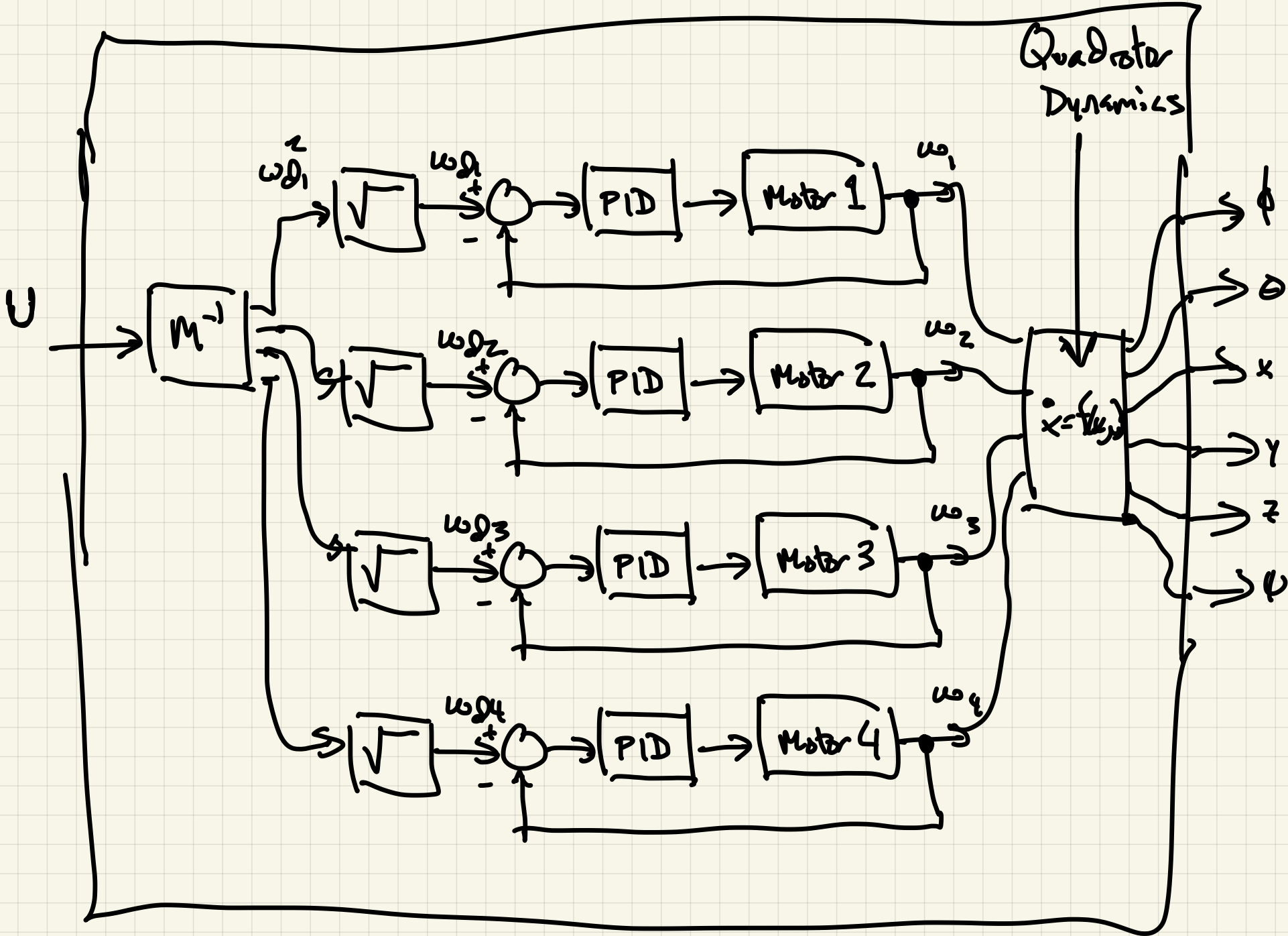
(d, h) pitch, tilt about y axis

$$U_2 = b(-\omega_2^2 + \omega_4^2)$$

$$U = \begin{bmatrix} U_1 \\ U_2 \\ U_3 \\ U_4 \end{bmatrix} = \begin{bmatrix} b & b & b & b \\ 0 & -b & 0 & b \\ -d & d & -d & d \\ -d & d & -d & d \end{bmatrix} \begin{bmatrix} \omega_1^2 \\ \omega_2^2 \\ \omega_3^2 \\ \omega_4^2 \end{bmatrix}$$

$$\Rightarrow \omega^2 = M^{-1} U$$

Quadrotor Dynamics



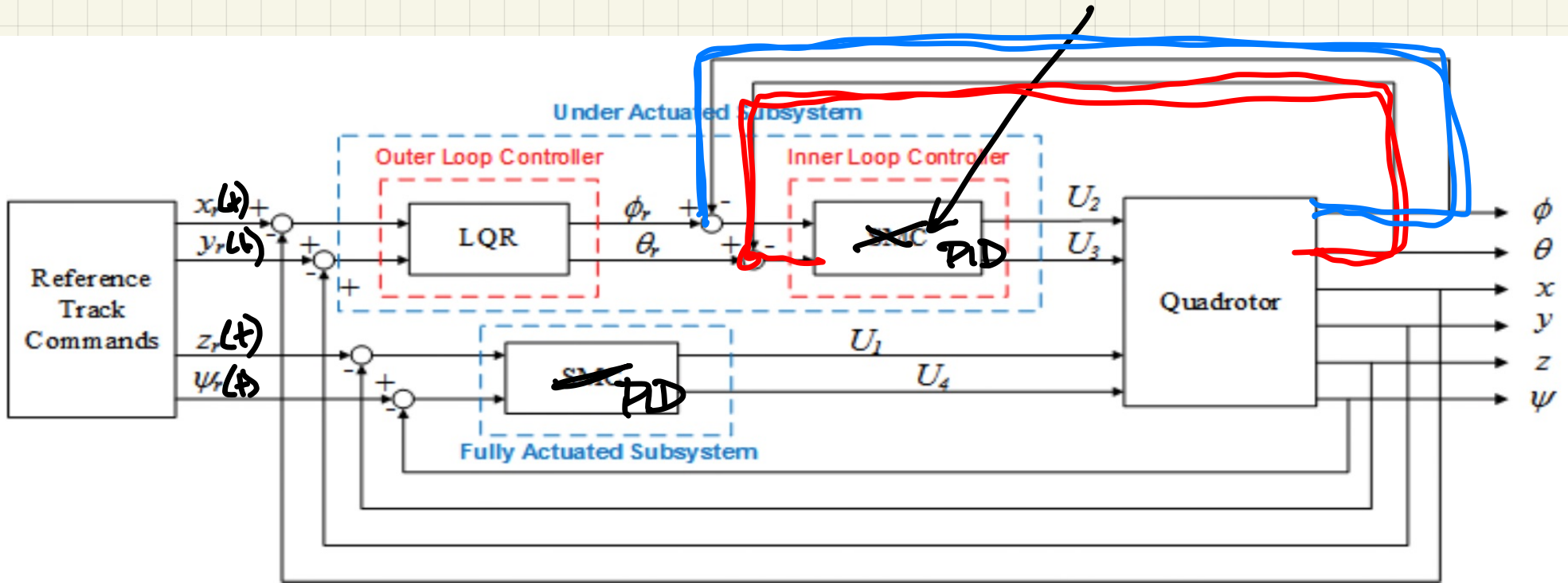
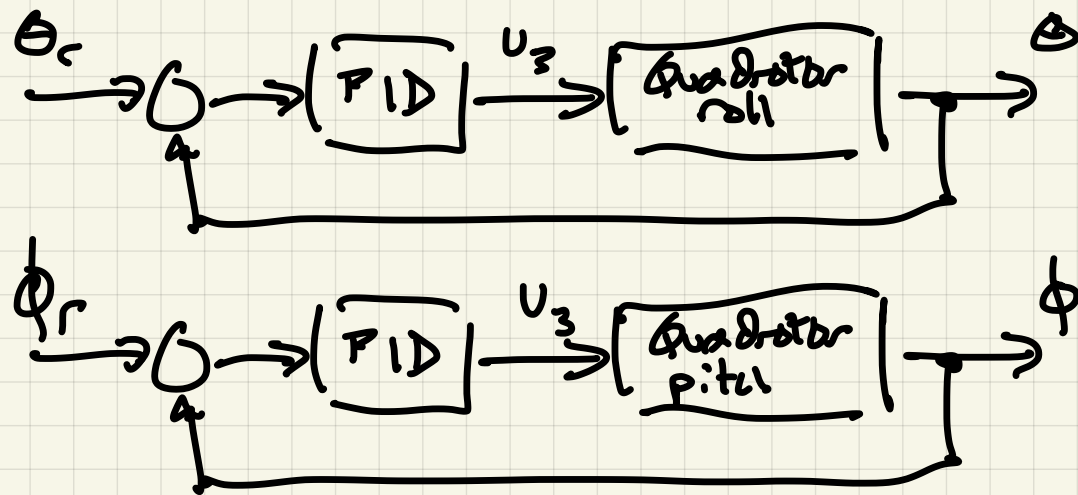


Fig. 2. UAV control system block diagram.

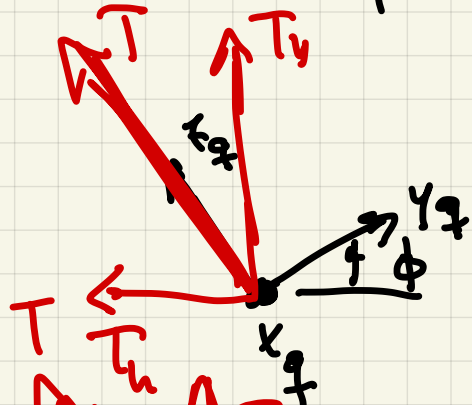
Ghamry, Khaled A., and Youmin Zhang. "Formation control of multiple quadrotors based on leader-follower method." In 2015 International Conference on Unmanned Aircraft Systems (ICUAS), pp. 1037-1042. IEEE, 2015.



Order Loop (LQR)

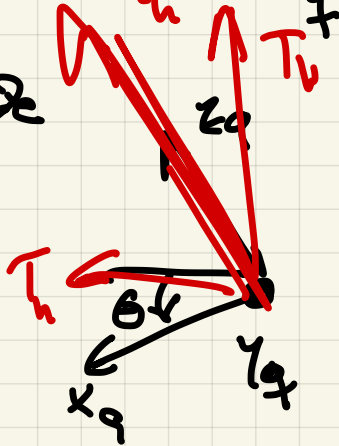
Two Things

① how $\phi, \theta \rightarrow x, y$
from front



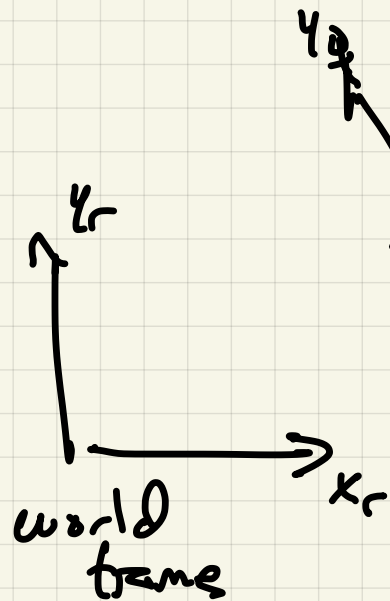
ϕ causes acceleration in the (sort of)
 $-y_q$ direction

from side



② cause acceleration in the (sort of)
 $+x_q$ direction

② x_r, y_r and x_q, y_q are not the same thing!



Rotation difference between r and q coordinates

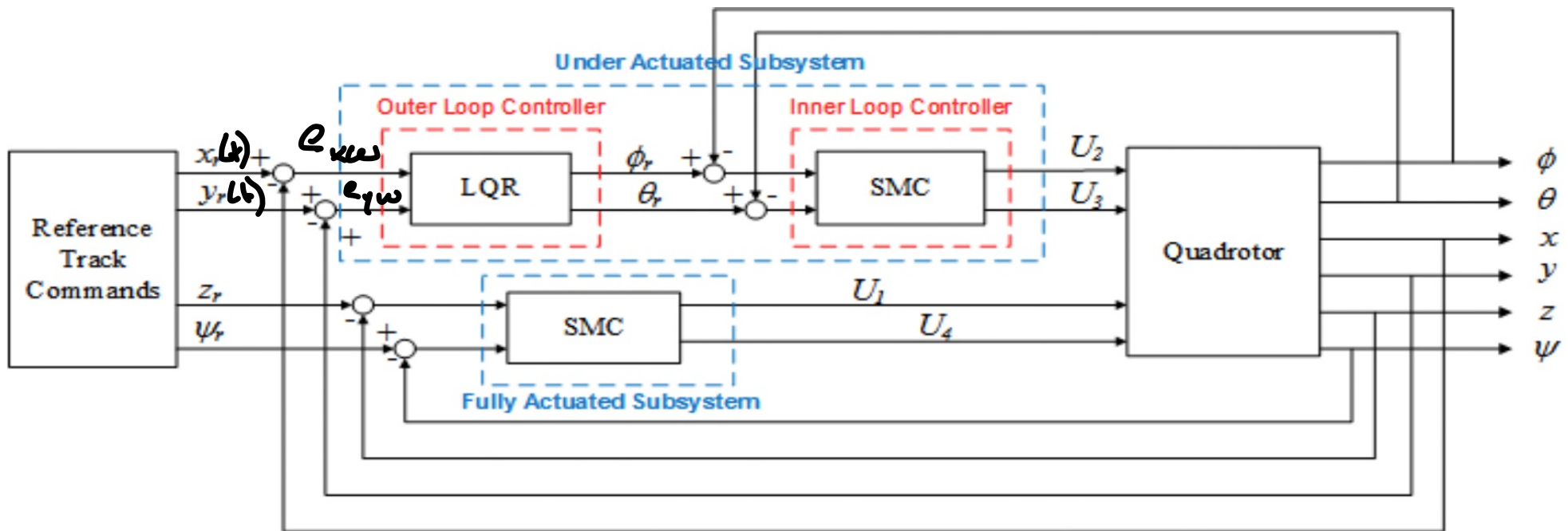
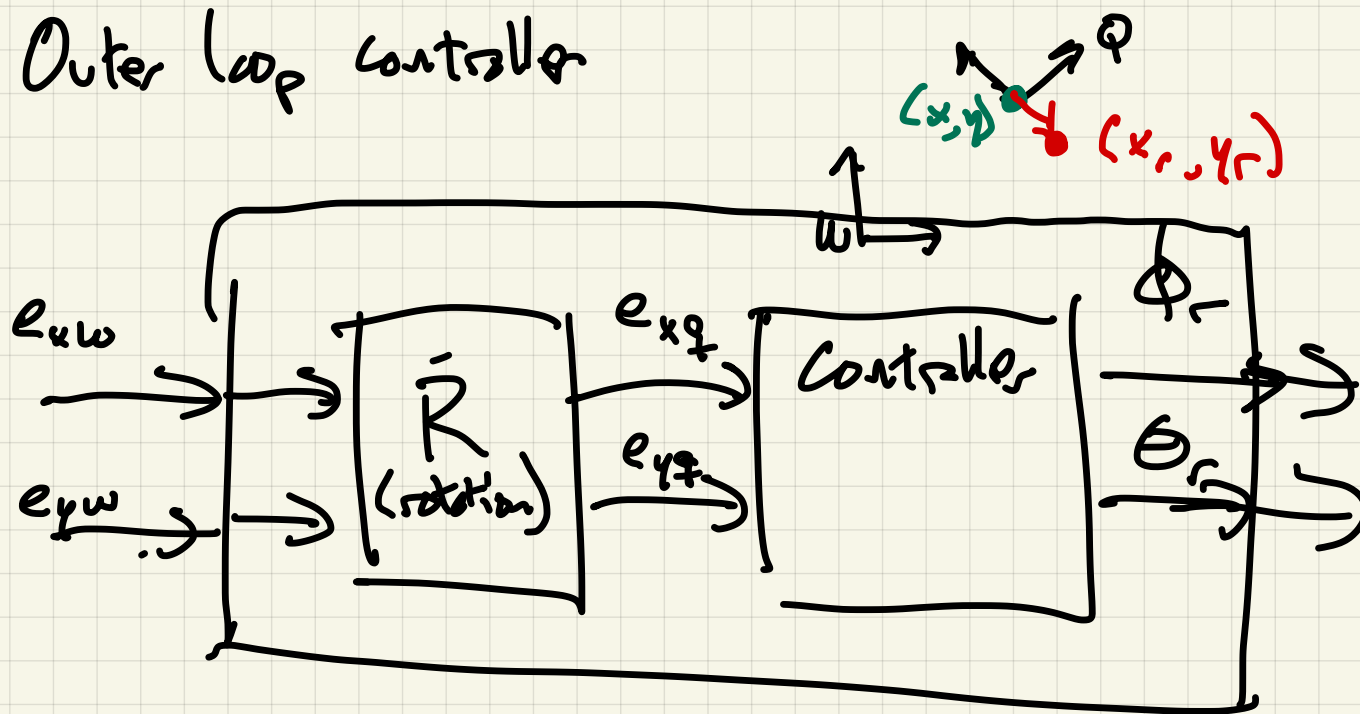


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Outer loop controller



3 Take Home Messages

(1) Loops within loops

(2) Decoupling (combination of intuition and math)

(3) control methods can be different